

In the 6EY18 engines, The problem that exhaust gas temperature won't be dropped although the turbocharger was disassembled and maintained occurs. The exhaust gas temperature rise is caused by faulty combustion. The major contributing factors for the faulty combustion are (1) inappropriate heating temperature of H.F.O., (2) faulty FO injection and (3) performance drop of turbocharger. In order to solve the problem, implement the corrective measures for each of the contributing factor as described below.

1. Inappropriate Heating Temperature of HFO (Inappropriate Viscosity of HFO)

When operating on HFO, the fuel temperature must be controlled so that the viscosity at the engine inlet becomes to be 11~14cSt. To obtain the appropriate heating temperature, refer to the example shown in the operation manual. When the temperature is obtained, include appropriate temp. allowance for the temp. drop in the piping and adjust the setting of the heater and the viscosity control equipment.

If the engine inlet temperature can't be improved with the adjustment of the setting, it is possible that the balance between the engine inlet pressure, P1, and the return pressure, P2, has collapsed, which has caused the fuel flow volume to the engine to be insufficient, (see Fig.1 below). In this case, adjust the respective pressure regulating valve to obtain the following pressure:

Pressure Setting of Hull-mount Pressure Regulating Valve, P2 ≥ Engine Inlet Pressure, P3+0.05MPa

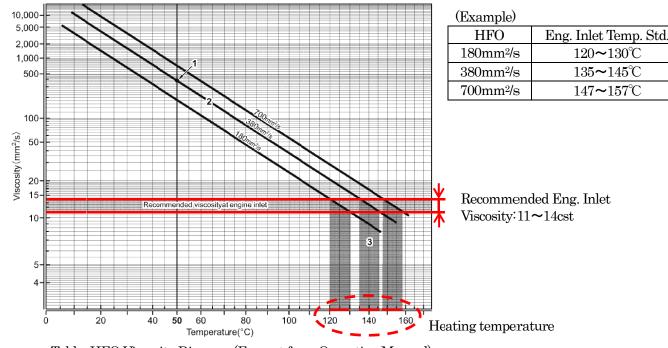


Table. HFO Viscosity Diagram (Excerpt from Operation Manual)

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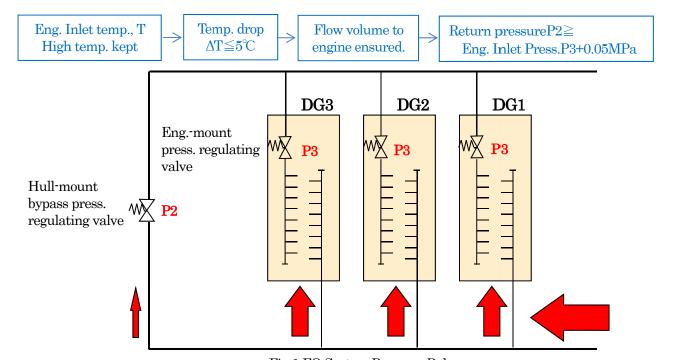


Fig.1 FO System Pressure Balance

2. Faulty FO Injection

Adjust the FO injection valve every $1500\sim2000$ hours, (first time, after $500\sim800$ hrs.) to ensure normal FO injection, (see instructions of the operation manual).

3. Turbocharger Performance Drop

(1) Turbine Wheel Contamination

Clean the turbine appropriately according to the inspections of the operation manual.)

(2)Increase of Clearance between Nozzle Ring and Exhaust Casing, (Clearance Q)
In the MET18 turbocharger, though after overhaul, exhaust gas temperature rose. The turbocharger manufacturer, Mitsubishi Heavy Industries Marine Machinery & Engine Co., Ltd, as a result of their survey on this problem, reported to us that the turbocharger performance dropped due to the increase of clearance between the nozzle ring and exhaust casing, (Clearance Q). Carbon was removed by the maintenance cleaning and the increased Q-clearance was exposed as a result. We project that this caused the exhaust temperature to rise soon after conducting maintenance.

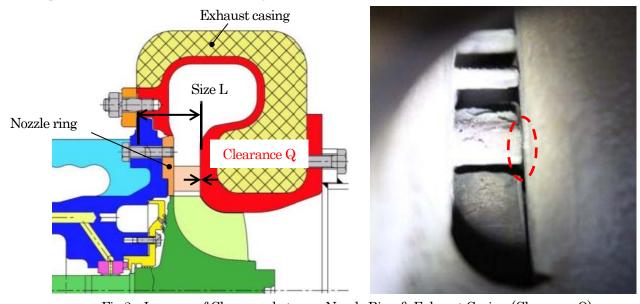


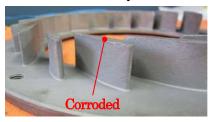
Fig.2 Increase of Clearance between Nozzle Ring & Exhaust Casing (Clearance Q)

①Cause of Trouble and Improvement

Cause

[Corrosion Wastage of Nozzle Ring]

Nozzle ring caused corrosion wastage due to high oxidation corrosion, which contributed to increase the Clearance Q.



[Nozzle ring rolled back]

As a result of FEM analysis, we found that the exhaust casing interfered with the nozzle ring outer circumference due to thermal expansion to cause distortion. However, this failure is corrected upon installing the exhaust casing. Accordingly, this factor gives only very small effect to the turbocharger performance drop.





[Distortion of Exhaust Casing]

We found that the distance between the installation flange face and the opposing nozzle face, (Size L in Fig.2), is distorted due to thermal stress for deformation, which contributed to increase the Q-clearance. However, the distortion was extremely small. Accordingly, this factor gives only very small effect to the turbocharger performance drop

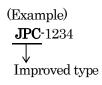
Improvement

[Nozzle Ring Material and External Diameter Change]

- ·In an aim to avoid interference caused by thermal expansion, the external diameter was diminished by 0.2mm.

[Discrimination]

Check the mfg. number punched on the nozzle ring surface and the rear casting surface.





Casting surface (4 positions)



[Applicable Engine No.]

 $7249 \sim 7251,7371 \sim 7385,7389 \sim 7403,7407 \sim 7409, \\7416 \sim 7419,7433 \sim 7441,7446 \sim 7448,7452 \sim 7454, \\7458 \sim 7466,7473 \sim 7490,7494 \sim$

[Exhaust Casing Material Change]

To increase strength against thermal stress, the material was changed as follows: $FCD450 \rightarrow Hi\text{-Si}$ ductile

(Discrimination)

Check the mfg. number punched on casing.

(Example) $15S\text{-}1257 \underset{}{\overset{}{\mathbf{H}}}$ Improved type



[Applicable Engine No.] 1840,1841,1855~

2Corrective Measure

As shown in the diagram below, it is possible to control the Q-clearance expansion by changing the present item to the improved nozzle ring. When the exhaust temperature rise is suspected due to Q-clearance expansion and performance drop, please inform to your Yanmar contact.

The exhaust casing needs not be replaced since the distortion was extremely small as mentioned above.

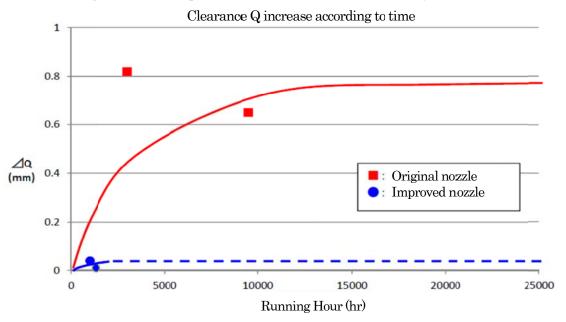


Table 2. Part Number of Improvement Nozzle Ring

Part No.	Engine Model	Output(kW)	Rotation(min ⁻¹)	Fuel Oil	
		800			
146621-19330 146621-19340		745			
	6EY18ALW	680		HFO, MDO	
		660	900, 1000		
		615	900, 1000		
		550			
		500			
		455			
146623-19330 146623-19340 146623-19350 146623-19360		800			
		745			
		680	900, 1000	HFO, MDO	
	6EY18AL	660			
		615			
		550			
		500	900		
		455			
		550			
		500	1000		
		455			
146625-19330		615			
146625-19340	6EY18L	550			
		500	720, 750	HFO, MDO	
		450			
146625-19350		400			

Attached Material: MET18 Turbocharger Performance Interview Sheet (TZ-E002-4653)

船名/Ship name 機関型式/Engine Model		機関番号/Engine Serial No.		過給機型式/Turbocharger Model			過給機製造番号/Turbocharger Serial No.			
	排気温度は整備後に高くなったものでしょうか?		□ YES					ı		
1	Did gas inlet temperature become higher after overhaul?		□NO							
	総運転時間、最後の開放点検日時とその際の運転時間を教示下さい。 Kindly report to MHI about Total running hour/ date of latest overhaul & the running hour at the time)		総運転時間/Total running hour: hr							
٦			<u> </u>							
2					h - 4'	日付 Date				
			Date of latest overhaul & the running hou					<u> </u>		hr
					就航直後 Just	就航6か月 6months	後	直近のデータ	清掃 Cle	
				/	after delivery	after delive	ry	Last data	前 Before	後 After
			日時 Date		-				Belore	7 (110)
	就航直後と、直近のデータを教示ください。 Kindly report to MHI about the operation data (Refer right table)		ガス入口温度						†	
			Gas inlet temp [degree]						<u> </u>	
3			ガス出口温度 Gas outlet temp[degree	5]					!	
			給気圧力	~]					+	
			Boosted air pressure [MI	Pa]						<u> </u>
			機関室温度							
			ENG room temp (suction air temp) [deg	ree1						
			エアクーラー冷却水温度	In					!	!
			Cooling water temp	Out					+	
<u> </u>		後の運転データも数三願います	for air cooler	Out						
4	整備をした実績があればその前後の運転データも教示願います。 If the actual result of overhaul maintenance, report to us the data(before & after overha		pefore & after overhaul)							J
5	直近の整備時の記録(隙間記録含)を教示願います。各隙間は許容値内でしょうか?		隙間 A,W, H, V, H を確認ください。							
Ľ	Kindly repot the latest overhaul record (with clearance table)/(Are they under our design or limit?)		Check the Clearance A,W, H, V, H							
6	直近の整備時の部品状況を写真にて教示願います。 Kindly send the latest photo of each parts (at overhaul) to MHI									
	主要部品(ノズル/ガスケーシング/羽根車/ディフューザ)の交換履歴があれば、その内容(日時含む)を 教示願います。									
7										
	If there is record of main part replacement (nozzle/gas casing/compabout detail with the replacement data.		mpressor /diffuser), kindly t	ell us						
	ガスケーシングと軸受台の間にある軟鋼パッキンが 2 枚重ねになっていないでしょうか Was the steel packing (p.No.309) used double sheets ? It leads to bigger tu				□ YES					
8				an	□ NO					
	was the steel packing (p.140	us to bigger turbine side go	aμ	•						